# **BYG140 KONSTRUKSJONSMEKANIKK 1**

# **Assignment (4) Solutions**

(Mechanics of Materials Ch 1: Stress, Ch 2: Strain, Ch 3: Mechanical Properties of Materials, Ch 4: Axial Load and Ch 6:Bending)

**Question 1** 

Determine the average normal stress at section a-a and the average shear stress at section b-b in member AB. The cross section is square, 12 mm on each side.

Consider the FBD of member BC, Fig. a,

$$\zeta + \Sigma M_C = 0$$
;  $F_{AB} \sin 60^{\circ} (1.2) - 2.5(1.2)(0.6) = 0$   $F_{AB} = 1.732 \text{ kN}$ 

Referring to the FBD in Fig. b,

$$^{+}\angle\Sigma F_{x'} = 0;$$
  $N_{a-a} + 1.732 = 0$   $N_{a-a} = -1.732 \text{ kN}$ 

$$N_{a-a} = -1.732 \text{ kN}$$

Referring to the FBD in Fig. c.

$$+\uparrow \Sigma F_v = 0;$$
  $V_{b-b} - 1.732 \sin 60^\circ = 0$   $V_{b-b} = 1.500 \text{ kN}$ 

$$V_{b-b} = 1.500 \text{ kN}$$

The cross-sectional areas of section a-a and b-b are  $A_{a-a}=12(12)=144$  mm<sup>2</sup> and

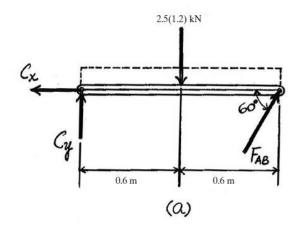
$$A_{b-b} = 12 \left( \frac{12}{\cos 60^{\circ}} \right) = 288 \text{ mm}^2$$
. Thus

$$\sigma_{a-a} = \frac{N_{a-a}}{A_{a-a}} = \frac{1.732(10^3)}{144} = 12.0 \text{ MPa}$$

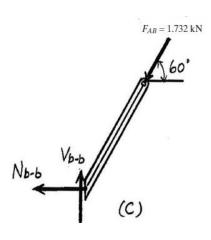
2.5 kN/m

$$\tau_{b-b} = \frac{V_{b-b}}{A_{b-b}} = \frac{1.500(10^3)}{288} = 5.21 \text{ MPa}$$

Ans.







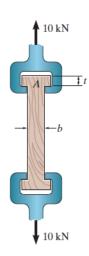
The wood specimen is subjected to the pull of 10 kN in a tension testing machine. If the allowable normal stress for the wood is  $(\sigma_t)_{\text{allow}} = 12 \text{ MPa}$  and the allowable shear stress is  $\tau_{\text{allow}} = 1.2 \text{ MPa}$ , determine the required dimensions b and t so that the specimen reaches these stresses simultaneously. The specimen has a width of 25 mm.

Allowable Shear Stress: Shear limitation

$$\tau_{\text{allow}} = \frac{V}{A};$$
  $1.2(10^6) = \frac{5.00(10^3)}{(0.025) t}$   $t = 0.1667 \text{ m} = 167 \text{ mm}$ 

Allowable Normal Stress: Tension limitation

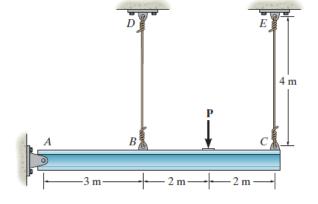
$$\sigma_{\text{allow}} = \frac{P}{A};$$
  $12.0(10^6) = \frac{10(10^3)}{(0.025) b}$   $b = 0.03333 \text{ m} = 33.3 \text{ mm}$ 

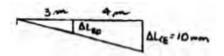




The rigid beam is supported by a pin at A and wires BD and CE. If the load  $\mathbf{P}$  on the beam causes the end C to be displaced 10 mm downward, determine the normal strain developed in wires CE and BD.

$$\begin{split} \frac{\Delta L_{BD}}{3} &= \frac{\Delta L_{CE}}{7} \\ \Delta L_{BD} &= \frac{3 \ (10)}{7} = 4.286 \ \mathrm{mm} \\ \varepsilon_{CE} &= \frac{\Delta L_{CE}}{L} = \frac{10}{4000} = 0.00250 \ \mathrm{mm/mm} \\ \varepsilon_{BD} &= \frac{\Delta L_{BD}}{L} = \frac{4.286}{4000} = 0.00107 \ \mathrm{mm/mm} \end{split}$$





## **Question 4**

The piece of plastic is originally rectangular. Determine the shear strain  $\gamma_{xy}$  at corners D and C if the plastic distorts as shown by the dashed lines.

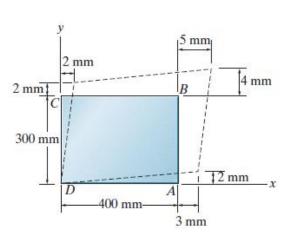
Geometry: For small angles,

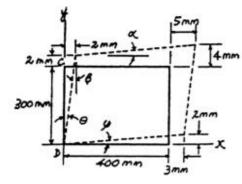
$$\alpha = \psi = \frac{2}{403} = 0.00496278 \,\text{rad}$$

$$\beta = \theta = \frac{2}{302} = 0.00662252 \text{ rad}$$

Shear Strain:

$$(\gamma_C)_{xy} = -(\alpha + \beta)$$
  
= -0.0116 rad = -11.6(10<sup>-3</sup>) rad  
 $(\gamma_D)_{xy} = \theta + \psi$   
= 0.0116 rad = 11.6(10<sup>-3</sup>) rad





The stress-strain diagram for a polyester resin is given in the figure. If the rigid beam is supported by a strut AB and post CD made from this material, determine the largest load P that can be applied to the beam before it ruptures. The diameter of the strut is 12 mm and the diameter of the post is 40 mm.

Given:

$$L_{AB} := 2m$$
  $L_{CD} := 0.5m$   $L_{AC} := 1.5m$   $d_{AB} := 12mm$   $d_{CD} := 40mm$ 

Solution:

$$A_y = \frac{P}{2} \qquad C_y = \frac{P}{2}$$

$$F_{AB} = A_y$$
  $F_{CD} = C_y$ 

$$\mathsf{Area}_{AB} := \left(\frac{\pi}{4}\right) \cdot \mathsf{d}_{AB}^2 \qquad \mathsf{Area}_{CD} := \left(\frac{\pi}{4}\right) \cdot \mathsf{d}_{CD}^2$$

### For rupture of strut AB:

From the stress-strain diagram,  $\sigma_{R_t} = 50.0 MPa$ 

$$F_{AB} = Area_{AB} \cdot (\sigma_{R_t})$$

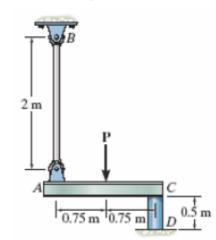
$$P := 2Area_{AB} \cdot (\sigma_{R_t})$$

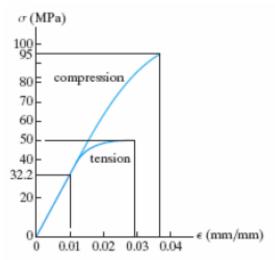
$$P = 11.31 \text{ kN}$$
(Controls!)

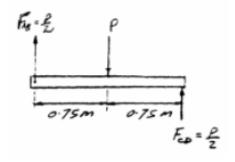
#### For rupture of post CD:

From the stress-strain diagram,  $\sigma_{R}$  c := 95.0MPa

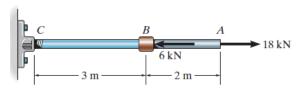
$$F_{CD}$$
 = Area<sub>CD</sub>· $(\sigma_{R_c})$   
 $P := 2Area_{CD}·(\sigma_{R_c})$   
 $P = 238.76 \text{ kN}$ 





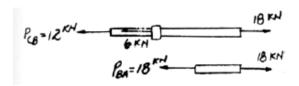


The assembly consists of a steel rod CB and an aluminum rod BA, each having a diameter of 12 mm. If the rod is subjected to the axial loadings at A and at the coupling B, determine the displacement of the coupling B and the end A. The unstretched length of each segment is shown in the figure. Neglect the size of the connections at B and C, and assume that they are rigid.  $E_{\rm st} = 200$  GPa,  $E_{\rm al} = 70$  GPa.



$$\delta_B = \frac{PL}{AE} = \frac{12(10^3)(3)}{\frac{\pi}{4}(0.012)^2(200)(10^9)} = 0.00159 \text{ m} = 1.59 \text{ mm}$$

$$\delta_A = \Sigma \frac{PL}{AE} = \frac{12(10^3)(3)}{\frac{\pi}{4}(0.012)^2(200)(10^9)} + \frac{18(10^3)(2)}{\frac{\pi}{4}(0.012)^2(70)(10^9)}$$



# **Question 7**

 $= 0.00614 \,\mathrm{m} = 6.14 \,\mathrm{mm}$ 

The three suspender bars are made of A992 steel and have equal cross-sectional areas of 450 mm<sup>2</sup>. Determine the average normal stress in each bar if the rigid beam is subjected to the loading shown.

Referring to the FBD of the rigid beam, Fig. a,

$$+\uparrow \Sigma F_{y} = 0;$$
  $F_{AD} + F_{BE} + F_{CF} - 50(10^{3}) - 80(10^{3}) = 0$ 

(3)

$$\zeta + \Sigma M_D = 0;$$
  $F_{RE}(2) + F_{CF}(4) - 50(10^3)(1) - 80(10^3)(3) = 0$  (2)

Referring to the geometry shown in Fig. b,

$$\delta_{BE} = \delta_{AD} + \left(\frac{\delta_{CF} - \delta_{AD}}{4}\right)(2)$$

$$\delta_{BE} = \frac{1}{2}\left(\delta_{AD} + \delta_{CF}\right)$$

$$\frac{F_{BE} \mathcal{L}}{\mathcal{A}\mathcal{E}} = \frac{1}{2}\left(\frac{F_{AD}\mathcal{L}}{\mathcal{A}\mathcal{E}} + \frac{F_{CF} \mathcal{L}}{\mathcal{A}\mathcal{E}}\right)$$

$$F_{AD} + F_{CF} = 2 F_{BE}$$

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Solving Eqs. (1), (2), and (3) yields

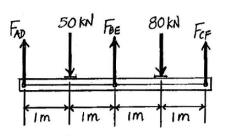
$$F_{BE} = 43.33(10^3) \text{ N}$$
  $F_{AD} = 35.83(10^3) \text{ N}$   $F_{CF} = 50.83(10^3) \text{ N}$ 

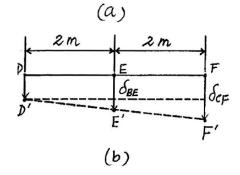
Thus,

$$\sigma_{BE} = \frac{F_{BE}}{A} = \frac{43.33(10^3)}{0.45(10^{-3})} = 96.3 \text{ MPa}$$

$$\sigma_{AD} = \frac{F_{AD}}{A} = \frac{35.83(10^3)}{0.45(10^{-3})} = 79.6 \text{ MPa}$$

$$\sigma_{CF} = 113 \text{ MPa}$$





If the material of the beam has an allowable bending stress of  $\sigma_{\text{allow}} = 150 \text{ MPa}$ , determine the maximum allowable intensity w of the uniform distributed load.

Support Reactions: As shown on the free-body diagram of the beam, Fig. a,

Maximum Moment: The maximum moment occurs when V = 0. Referring to the free-body diagram of the beam segment shown in Fig. b,

$$+\uparrow \sum F_{y} = 0;$$
  $3w - wx = 0$ 

$$x = 3 \,\mathrm{m}$$

$$\zeta + \sum M = 0;$$
  $M_{\text{max}} + w(3) \left(\frac{3}{2}\right) - 3w(3) = 0$   $M_{\text{max}} = \frac{9}{2}w$ 

$$M_{\text{max}} = \frac{9}{2}w$$

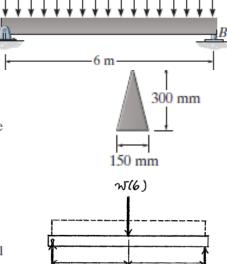
Ans.

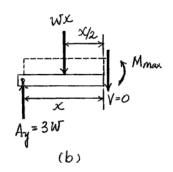
Section Properties: The moment of inertia of the cross section about the neutral axis is

$$I = \frac{1}{36}(0.15)(0.3^3) = 0.1125(10^{-3}) \,\mathrm{m}^4$$

**Absolute Maximum Bending Stress:** Here,  $c = \frac{2}{3}(0.3) = 0.2 \text{ m}.$ 

$$\sigma_{\text{allow}} = \frac{Mc}{I}$$
; 
$$150(10^6) = \frac{\frac{9}{2}w(0.2)}{0.1125(10^{-3})}$$
$$w = 18750 \text{ N/m} = 18.75 \text{ kN/m}$$





The box beam is subjected to the internal moment of  $M = 4 \text{ kN} \cdot \text{m}$ , which is directed as shown. Determine the maximum bending stress developed in the beam and the orientation of the neutral axis.

**Internal Moment Components:** The y component of M is negative since it is directed towards the negative sense of the y axis, whereas the z component of M which is directed towards the positive sense of the z axis is positive, Fig. a. Thus,

$$M_y = -4 \sin 45^\circ = -2.828 \text{ kN} \cdot \text{m}$$
  
 $M_z = 4 \cos 45^\circ = 2.828 \text{ kN} \cdot \text{m}$ 

**Section Properties:** The moments of inertia of the cross section about the principal centroidal y and z axes are

$$I_y = \frac{1}{12}(0.3)(0.15^3) - \frac{1}{12}(0.2)(0.1^3) = 67.7083(10^{-6}) \text{ m}^4$$
  
 $I_z = \frac{1}{12}(0.15)(0.3^3) - \frac{1}{12}(0.1)(0.2^3) = 0.2708(10^{-3}) \text{ m}^4$ 

Bending Stress: By inspection, the maximum bending stress occurs at corners A and D.

$$\sigma = -\frac{M_z y}{I_z} + \frac{M_y z}{I_y}$$

$$\sigma_{\text{max}} = \sigma_A = -\frac{2.828(10^3)(0.15)}{0.2708(10^{-3})} + \frac{(-2.828)(10^3)(0.075)}{67.7083(10^{-6})}$$

$$= -4.70 \text{ MPa} = 4.70 \text{ MPa}(C)$$

$$\sigma_{\text{max}} = \sigma_D = -\frac{2.828(10^3)(-0.15)}{0.2708(10^{-3})} + \frac{(-2.828)(10^3)(-0.075)}{67.7083(10^{-6})}$$

$$= 4.70 \text{ MPa}(T)$$
Ans.

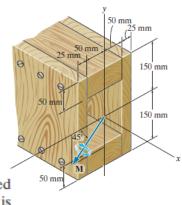
Orientation of Neutral Axis: Here,  $\theta = -45^{\circ}$ .

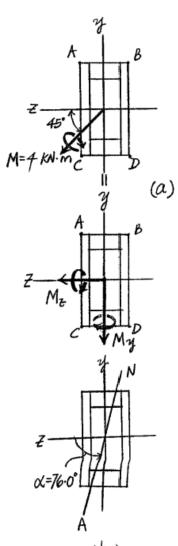
$$\tan \alpha = \frac{I_z}{I_y} \tan \theta$$

$$\tan \alpha = \frac{0.2708(10^{-3})}{67.7083(10^{-6})} \tan (-45^\circ)$$

$$\alpha = -76.0^\circ$$
Ans.

The orientation of the neutral axis is shown in Fig. b.





A wood beam is reinforced with steel straps at its top and bottom as shown. Determine the maximum bending stress developed in the wood and steel if the beam is subjected to a bending moment of  $M=5~\mathrm{kN}\cdot\mathrm{m}$ . Sketch the stress distribution acting over the cross section. Take  $E_\mathrm{w}=11~\mathrm{GPa}, E_\mathrm{st}=200~\mathrm{GPa}$ .

$$n = \frac{200}{11} = 18.182$$

$$I = \frac{1}{12}(3.63636)(0.34)^3 - \frac{1}{12}(3.43636)(0.3)^3 = 4.17848(10^{-3}) \text{ m}^4$$

Maximum stress in steel:

$$(\sigma_{\rm st})_{\rm max} = \frac{nMc_1}{I} = \frac{18.182(5)(10^3)(0.17)}{4.17848(10^{-3})} = 3.70 \text{ MPa}$$

Maximum stress in wood:

$$(\sigma_{\rm w})_{\rm max} = \frac{Mc_2}{I} = \frac{5(10^3)(0.15)}{4.17848(10^{-3})} = 0.179 \text{ MPa}$$

$$(\sigma_{st}) = n(\sigma_{w})_{max} = 18.182(0.179) = 3.26 \text{ MPa}$$

